Earth Science Applications of Space Based Geodesy DES-7355 Tu-Th 9:40-11:05

Seminar Room in 3892 Central Ave. (Long building)

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http://www.ceri.memphis.edu/people/smalley/ESCI7355/ESCI_7355_Applications_of_Space_Based_Geodesy.html



Using double difference phase observations for relative positioning

First notice that if we make all double differences - even ignoring the obvious duplications

$$\nabla \Delta L_{AB}^{jk} = \nabla \Delta L_{AB}^{kj} = \nabla \Delta L_{BA}^{kj} = \nabla \Delta L_{BA}^{jk}$$

We get a lot more double differences than original data. This can't be (can't create information).

Consider the case of 3 satellites observed by 2 receivers.

Form the (non trivial) double differences

$$L_{AB}^{jk} = \left(L_A^j - L_B^j\right) - \left(L_A^k - L_B^k\right)$$
$$L_{AB}^{jl} = \left(L_A^j - L_B^j\right) - \left(L_A^l - L_B^l\right)$$
$$L_{AB}^{lk} = \left(L_A^l - L_B^l\right) - \left(L_A^k - L_B^k\right)$$



Note that we can form any one from a linear combination of the other two

 $L_{AB}^{jk} = L_{AB}^{jl} - L_{AB}^{lk}$ $L_{AB}^{jl} = L_{AB}^{jk} - L_{AB}^{lk}$ $L^{lk}_{AB} = L^{jk}_{AB} - L^{jl}_{AB}$

(línearly dependent) We need a linearly independent set for Least Squares.

From the linearly dependent set

$$\left\{L_{AB}^{jk},\!L_{AB}^{jl},\!L_{AB}^{lk}
ight\}$$

We can form a number of linearly independent subsets

$$\left\{ L_{AB}^{jk}, L_{AB}^{jl} \right\} = \Lambda^{j} = \left\{ L_{AB}^{ab} \middle| a = j; b \neq j \right\}$$
$$\left\{ L_{AB}^{kj}, L_{AB}^{kl} \right\} = \Lambda^{k} = \left\{ L_{AB}^{ab} \middle| a = k; b \neq k \right\}$$
$$\left\{ L_{AB}^{lj}, L_{AB}^{lk} \right\} = \Lambda^{l} = \left\{ L_{AB}^{ab} \middle| a = l; b \neq l \right\}$$

Which we can then use for our Least Squares estimation.

How to pick the basis?

All linearly independent sets are "equally" valid and should produce identical solutions. Pick A^I such that <u>reference satellite</u> / has data at every epoch

Better (but harder) approach is to select the reference satellite epoch by epoch

(if you have 24 hour data file, cannot pick one satellite and use all day - no satellite is visible all day)

For a single baseline (2 receivers) that observe *s* satellites, the number of linearly independent double difference observations is

5-1

Blewitt, Basics of GPS in "Geodetic Applications of GPS"

Next suppose we have more than 2 receivers. We have the same situation

-all the double differences are not linearly independent.

As we just did for multiple satellites, we can pick a reference station

that is common to all the double differences.

For a network of *r* receivers, the number of linearly independent double difference observations is

r-1

So all together we have a total of (s-1) (r-1) Linearly independent double differences

So our linearly independent set of double differences is

$$\Lambda_{C}^{j} = \left\{ L_{AB}^{ab} \middle| a = j; b \neq j; A = C, B \neq C \right\}$$

Blewitt, Basics of GPS in "Geodetic Applications of GPS"

Reference station method has problems when all receivers can't see all satellites at the same time.

Choose receiver close to center of network.

Even this might not work when the stations are very far apart.

For large networks may have to pick short baselines that connect the entire network.

Idea is to not have any closed polygons (which give multiple paths and are therefore linearly dependent) in the network.

Can also pick the reference station epoch per epoch.

If all the receivers see the same satellites at each epoch,

and data weighting is done properly,

then it does not matter which receiver and satellite we pick for the reference.

Blewitt, Basics of GPS in "Geodetic Applications of GPS"

In practice, however,

the solution depends on our choices of reference receiver and satellite.

(although the solutions should be similar)

(could process all undifferenced phase observatons and estimate clocks at each epoch - ideally gives "better" estimates) Double difference observation equations

Start with

 $\nabla \Delta L_{AB}^{jk} = \nabla \Delta \rho_{AB}^{jk} + \nabla \Delta Z_{AB}^{jk} - \nabla \Delta I_{AB}^{jk} - \nabla \Delta N_{AB}^{jk}$

Simplify to $L_{AB}^{jk} = \rho_{AB}^{jk} - \lambda_0 N_{AB}^{jk}$

By dropping the $\nabla \Delta$ And assuming $\nabla \Delta Z_{AB}^{jk} \& \nabla \Delta I_{AB}^{jk}$ are negligible Processing double differences between two receivers results in a

Baseline solution

The estimated parameters include the vector between the two receivers (actually antenna phase centers). May also include estimates of parameters to model troposphere (statistical) and ionosphere (measured – dispersion). Also have to estimate the

Integer Ambiguities

For each set of satellite-receiver double differences

Blewitt, Basics of GPS in "Geodetic Applications of GPS"

We are faced with the same task we had before when we used

pseudo range

We have to

linearize

the problem in terms of the parameters we want to estimate

A significant difference between using the pseudo range, which is a stand alone method, and using the Phase, is that the phase is a <u>differential</u> method (similar to VLBI).



So far we have cast the problem in terms of the distances to the satellites, but we could recast it in terms of the relative distances between stations.



So now we will need multiple receivers. We will also have to use (at least one) as a reference station. In addition to knowing where the satellites are,

In addition to knowing where the satellites are, We need to know the position of the refrence station(s) to the same level of precision as we wish to estimate the position of the other stations.



fiducial positioning

Fiducial

Regarded or employed as a standard of reference, as in surveying.

http://dictionary.reference.com/search?q=fiducial

So now we have to assign the location of our fiducial station(s)

Can do this with

RINEX header position VLBI position

Other GPS processing

etc.

So we have to

Write down the equations Linearize Solve

Blewitt, Basics of GPS in "Geodetic Applications of GPS"

Double difference observation equations

Start with

 $\nabla \Delta L_{AB}^{jk} = \nabla \Delta \rho_{AB}^{jk} + \nabla \Delta Z_{AB}^{jk} - \nabla \Delta I_{AB}^{jk} - \nabla \Delta N_{AB}^{jk}$

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So we have to

Write down the equations Linearize Solve

Let the "reference" (also KNOWN) station be A

We want to estimate (x_B, y_B, z_B)

Using observations of satellites 1, 2, 3, and 4 (common observations at all epochs)

We also need to pick a "reference" satellite (position of all satellites known) Pick satellite 2.

(we have to pick the reference station and satellite to properly form a <u>linearly independent set</u> of double differences)

For each epoch *i*

We have the following 3 linearly independent sets of double difference observations

$$\Lambda_A^2(i) = \left\{ L_{AB}^{ab}(i) \middle| a = 2; b \neq 2 \right\}$$

$$\Lambda_{A}^{2}(i) = \left\{ L_{AB}^{21}(i), L_{AB}^{23}(i), L_{AB}^{24}(i) \right\}$$

To estimate the parameter set $\left\{x_{B}, y_{B}, z_{B}, N_{AB}^{21}, N_{AB}^{23}, N_{AB}^{24}\right\}$

(if there were no cycle slips, else we would have to estimate additional $N_{AB}^{ij}(k)$ term for each cycle slip, k.

As before, the linearized observation equations can be written in terms of the "usual suspects"

b = Ax + v

Residuals - dx1

Observation errors -dx1

"design" matrix – dxp

Parameter corrections – $p \times 1$ d – number linearly independent observables p – number of parameters to estimate Blewitt, Basics of GPS in "Geodetic Applications of GPS" In comparison to the pseudo range data, where we assumed the errors in the observables were independent,

the errors in double differenced data are not - the errors are correlated.

This means that we should use Weighted Least Squares The WLS solution to the normal equations is

$$\hat{x} = \left(A^T W A\right)^{-1} A^T W \vec{b}$$

Where W is (an appropriately formed) data weight matrix.

The covariance matrix is now given by (does this look familiar?) $C_x = \left(A^T W A\right)^{-1}$ The covariance matrix now has information about both the geometry (as before) And new (information or effects due to) correlations between the observables.

(if we assume, as for pseudo range, that the error in measurement of the phase is the same for all measurements – we can factor out a σ ,

But the differencing introduces a correlation between the "independent" measurements that makes the errors "leak" from one observable to another)

Again, one can get important information from the Covariance matrix

If it is not invertable mathematically (linearly dependent)

If it is not invertable practically/numerically (almost linearly dependent, large condition number)

Practically, can tell if all the integer ambiguities can be fixed. If so, get statistically better estimations.

Coefficients of the design matrix Look at one row.

$$\Lambda_{AB}^{24}(i) = \left\{ \frac{\partial L_{AB}^{24}(i)}{\partial x_B}, \frac{\partial L_{AB}^{24}(i)}{\partial y_B}, \frac{\partial L_{AB}^{24}(i)}{\partial z_B}, \frac{\partial L_{AB}^{24}(i)}{\partial N_{AB}^{21}}, \frac{\partial L_{AB}^{24}(i)}{\partial N_{AB}^{23}}, \frac{\partial L_{AB}^{24}(i)}{\partial N_{AB}^{24}} \right\}$$

$$L_{AB}^{jk} = \rho_{AB}^{jk} - \lambda_0 N_{AB}^{jk}$$

$$\Lambda_{AB}^{24}(i) = \left\{ \frac{\partial \rho_{AB}^{24}(i)}{\partial x_B}, \frac{\partial \rho_{AB}^{24}(i)}{\partial y_B}, \frac{\partial \rho_{AB}^{24}(i)}{\partial z_B}, 0, 0, -\lambda_0 \right\}$$

Blewitt, Basics of GPS in "Geodetic Applications of GPS"

Coefficients of the design matrix
Look at one derivative.

$$\frac{\partial \rho_{AB}^{24}(i)}{\partial x_{B}} = \frac{\partial}{\partial x_{B}} \left(\rho_{A}^{2}(i) - \rho_{B}^{2}(i) - \rho_{A}^{4}(i) + \rho_{B}^{4}(i) \right)$$

$$\frac{\partial \rho_{AB}^{24}(i)}{\partial x_{B}} = \frac{\partial \rho_{A}^{2}(i)}{\partial x_{B}} - \frac{\partial \rho_{B}^{2}(i)}{\partial x_{B}} - \frac{\partial \rho_{A}^{4}(i)}{\partial x_{B}} + \frac{\partial \rho_{B}^{4}(i)}{\partial x_{B}}$$
Independent of x_{B}

$$\frac{\partial \rho_{AB}^{24}(i)}{\partial x_{B}} = \frac{\partial \rho_{B}^{4}(i)}{\partial x_{B}} - \frac{\partial \rho_{B}^{2}(i)}{\partial x_{B}}$$

$$\frac{\partial \rho_{AB}^{24}(i)}{\partial x_{B}} = \frac{\partial \rho_{B}^{4}(i)}{\partial x_{B}} - \frac{\partial \rho_{B}^{2}(i)}{\partial x_{B}}$$

Coefficients of the design matrix Finally one can use the relationship between Range and Time and Time and Phase (what we measured). $\rho_A^j(i) = c \left(T_A(i) - T^j(i) \right)$

$$\phi(T) = f_0 T + \phi_0$$

To write everything in terms of the observables.

Final detail

Minimum data requirements

Necessary (but not sufficient condition) that

Number of data Exceed

Number of parameters to estimate.

So we have

d≥p (allowing perfect solution d=p) If all receivers track the same satellites there are d=q(r-1)(s-1)Linearly independent double differences Where q is the number of epochs r the number of receivers s the number of satellites

Assuming no cycle slips p=3+(r-1)(s-1)So $d=q(r-1)(s-1) \ge 3+(r-1)(s-1)$

 $(q-1)(r-1)(s-1) \ge 3$

So for *r=2*, *s=2*

 $q \ge 4$ (gives one double difference per epoch)

Common-mode Cancellations

Observation	Effects elíminated	Effects reduced	Option
Single dífferences.	Satellite <u>or</u> station clock (first order).	Orbit errors. GDOP. ionosphere	Constraín ambiguíty.
Double dífferences.	Satellite <u>and</u> station clock (first order).	Orbit errors. GDOP. Ionosphere.	Constraín ambiguíty.
Triple dífferences.	Satellite <u>and</u> station clock (first order).		Ambiguity eliminated. Find-fix cycle slips

RINEX files Receiver Independent Exchange files (standard GPS, now GNSS, observables - data - file) ASCII files (text - you can read them) New competitor - may replace RINEX -BINEX Binary Exchange files (binary - can't read files without program, much more general == complicated)

RINEX Files have two basic parts



Data (observables)

TABLE A1 GPS OBSERVATION DATA FILE - HEADER SECTION DESCRIPTION			
HEADER LABEL (Columns 61-80)	HEADER LABEL DESCRIPTION (Columns 61-80)		
RINEX VERSION / TYPE - Format version (2.10) - File type ('0' for Observation Data) - Satellite System: blank or 'G': GPS 'R': GLONASS 'S': Geostationary signal payload 'T': NNSS Transit 'M': Mixed		F9.2,11X, A1,19X, A1,19X A1,19X	
PGM / RUN BY / DATE	 Name of program creating current file Name of agency creating current file Date of file creation 	A20, A20, A20	
+ * COMMENT +	Comment line(s)	A60	
MARKER NAME	Name of antenna marker	A60	
* MARKER NUMBER	Number of antenna marker	A20	
OBSERVER / AGENCY	Name of observer / agency	A20,A40	

REC # / TYPE / VERS	Receiver number, type, and version (Version: e.g. Internal Software Version)	3A20
ANT # / TYPE	Antenna number and type	2A20
APPROX POSITION XYZ	Approximate marker position (WGS84)	3F14.4
ANTENNA: DELTA H/E/N	 Antenna height: Height of bottom surface of antenna above marker Eccentricities of antenna center relative to marker to the east and north (all units in meters) 	3F14.4
WAVELENGTH FACT L1/2	 Default wavelength factors for L1 and L2 1: Full cycle ambiguities 2: Half cycle ambiguities (squaring) 0 (in L2): Single frequency instrument 	216,
	- zero or blank The default wavelength factor line is required and must preceed satellite- specific lines.	IG

* WAVELENGTH FACT L1/2	- Wavelength factors for L1 and L2 1: Full cycle ambiguities	216,	- *
	 2: Half cycle ambiguities (squaring) 0 (in L2): Single frequency instrument Number of satellites to follow in list for which these factors are valid. List of PRNs (satellite numbers with system identifier) 	I6, 7(3X,A1,I2)	
	These opional satellite specific lines may follow, if they identify a state different from the default values. Repeat record if necessary.		

# / TYPES OF OBSERV 	 Number of different observation types stored in the file Observation types 	I6, 9(4X,A2)
	If more than 9 observation types:	6X 9(4X A2)
	<pre>Dise continuation line(s) The following observation types are defined in RINEX Version 2.10: L1, L2: Phase measurements on L1 and L2 C1</pre>	6x,9(4x,Az)
	values as given by the receiver for the L1,L2 phase observations	
	are converted to "L2" or "P2" and flagged with bit 2 of loss of lock indicator (see Table A2).	

Units : Phase : full cycles
Pseudorange : meters
Doppler : Hz
Transit : cycles
SNR etc : receiver-dependent
The sequence of the types in this record
has to correspond to the sequence of the
observations in the observation records

				L
*	INTERVAL	Observation interval in seconds	F10.3	- * +
	TIME OF FIRST OBS	 Time of first observation record (4-digit-year, month,day,hour,min,sec) Time system: GPS (=GPS time system) GLO (=UTC time system) Compulsory in mixed GPS/GLONASS files Defaults: GPS for pure GPS files GLO for pure GLONASS files 	5I6,F13.7, 5X,A3	
*	TIME OF LAST OBS	 Time of last observation record (4-digit-year, month,day,hour,min,sec) Time system: Same value as in TIME OF FIRST OBS record 	5I6,F13.7, 5X,A3	*
*	RCV CLOCK OFFS APPL	Epoch, code, and phase are corrected by applying the realtime-derived receiver clock offset: 1=yes, 0=no; default: 0=no Record required if clock offsets are reported in the EPOCH/SAT records	I6	*
*	LEAP SECONDS	Number of leap seconds since 6-Jan-1980 Recommended for mixed GPS/GLONASS files	I6	* +
*	# OF SATELLITES	Number of satellites, for which observations are stored in the file	I6	*

* PRN / # OF OBS	<pre>PRN (sat.number), number of observations for each observation type indicated in the "# / TYPES OF OBSERV" - record. If more than 9 observation types: Use continuation line(s) This record is (these records are) repeated for each satellite present in the data file</pre>	3X,A1,I2,9I6
++ END OF HEADER ++	Last record in the header section.	++ 60X ++

Records marked with * are optional

Header example

2.10 OBSERVATION DATA M (MIXED) BLANK OR G = GPS, R = GLONASS, T = TRANSIT, M = MIXED XXRINEXO V9.9 AIUB 24-MAR-01 14:43 EXAMPLE OF A MIXED RINEX FILE A 9080 9080.1.34 BILL SMITH ABC INSTITUTE X1234A123 XX ZZZ 234 YY 4375274. 587466. 4589095. .9030 .0000 .0000 1 1 1 2 6 G14 G15 G16 G17 G18 G19 0 4 L1 L2 P2 P1 18,000 24 13 10 36,000000 2001 3

RINEX VERSION / TYPE COMMENT PGM / RUN BY / DATE COMMENT MARKER NAME MARKER NUMBER **OBSERVER** / AGENCY **REC # / TYPE / VERS** ANT # / TYPE APPROX POSITION XYZ ANTENNA: DELTA H/E/N WAVELENGTH FACT L1/2 WAVELENGTH FACT L1/2 RCV CLOCK OFFS APPL **# / TYPES OF OBSERV** INTERVAL TIME OF FIRST OBS END OF HEADER

(I've not seen many headers with the "time of last observation" line)

Another header example

2.10 OBSERVATION DATA G (GPS) RINEX VERSION / TYPE tegc 2005Feb10 You don't know? 20050411 15:07:57UTCPGM / RUN BY / DATE Linux 2.0.36 Pentium II gcc Linux 486/DX+ COMMENT BIT 2 OF LLIGFLAGS DATA COLLECTED UNDER A/S CONDITION COMMENT MARKER NAME CJTR OBSERVER / A ENCY -Unknown--Unknown-664 ASHTECH Z-12 CD00 REC # / TYPE / VERS 943 -Unknown-ANT # / TYPE 0.0000 0.0000 0.0000 APPROX POSITION XYZ 0.0000 0.0000 0.0000 ANTENNA: DELTA H/E/N WAVELENGTH FACT L1/2 1 1 5 L1 L2 C1 P1 P2 # / TYPES OF OBSERV SNR is mapped to RINEX snr flag value [0-9] COMMENT L1 & L2: 2-19 dBHz = 1, 20-27 dBHz = 2, 28-31 dBHz = 3 COMMENT 32-35 dBHz = 4, 36-38 dBHz = 5, 39-41 dBHz = 6COMMENT 42-44 dBHz = 7, 45-48 dBHz = 8, >= 49 dBHz = 9 COMMENT pseudorange smoothing corrections not applied COMMENT 2004 12 26 30.0000000 0 0 TIME OF FIRST OBS END OF HEADER

Not having an XO estimate makes processing more difficult

RINEX Observations (data)

TABLE A2 GPS OBSERVATION DATA FILE - DATA RECORD DESCRIPTION			
OBS. RECORD	DESCRIPTION	FORMAT	
EPOCH/SAT or EVENT FLAG	<pre>- Epoch : - year (2 digits, padded with 0 if necessary) - month,day,hour,min, - sec</pre>	1X,I2.2, 4(1X,I2), F11.7,	
	2X,I1,		
	 Number of satellites in current epoch List of PRNs (sat.numbers with system identifier, see 5.1) in current epoch 		
	 receiver clock offset (seconds, optional) If more than 12 satellites: Use continuation line(s) 	F12.9 32X, 12(A1,I2)	
	If epoch flag 2-5:		

01 3 24 13 10	36.0000000 0	3G12G 9G 6		123456789
23629347.915	.300	8353	23629364.158	
20891534.648	120	9358	20891541.292	
20607600.189	430	9.394	20607605.848	
01 3 24 13 10	54.0000000 0	5G12G 9G 6R21R22		123456789
23619095.450	-53875.632	8 -41981.375	23619112.008	
20886075.667	-28688.027	9 -22354.535	20886082.101	
20611072.689	18247.789	9 14219.770	20611078.410	
21345678.576	12345.567	5		
22123456.789	23456.789	5		

Notice the order of satellites, and which satellites are recorded is different for each epoch

RINEX Observations (data)

- Event flag:	[2X,I1,]
2: start moving antenna	
3: new site occupation (end of kinem. data)	
(at least MARKER NAME record follows)	
4: header information follows	
5: external event (epoch is significant,	
same time frame as observation time tags)	
- "Number of satellites" contains number of	[13]
special records to follow.	
Maximum number of records: 999	
- For events without significant epoch the	
epoch fields can be left blank	
If epoch flag = 6:	
6: cycle slip records follow to optionally	
report detected and repaired cycle slips	
(same format as OBSERVATIONS records:	
slip instead of observation: LLT and	
signal strength blank or zero)	

```
OBSERVATIONS - Observation
                                    rep. within record for
                                                                  m(F14.3,
               - LLI
                                   each obs.type (same seq
                                                                     I1,
               - Signal strength | as given in header)
                                                                     I1)
               If more than 5 observation types (=80 char):
               continue observations in next record.
               This record is (these records are) repeated for
               each satellite given in EPOCH/SAT - record.
               Observations:
                 Phase : Units in whole cycles of carrier
                 Code : Units in meters
               Missing observations are written as 0.0
               or blanks
               Phase values overflowing the fixed format F14.3
               have to be clipped into the valid interval (e.g.
               add or subtract 10**9), set LLI indicator.
               Loss of lock indicator (LLI). Range: 0-7
                0 or blank: OK or not known
                Bit 0 set : Lost lock between previous and
                            current observation: cycle slip
                            possible
                Bit 1 set : Opposite wavelength factor to the
                            one defined for the satellite by a
                            previous WAVELENGTH FACT L1/2 line.
                           Valid for the current epoch only.
```

RINEX Observations (data)

Bit 2 set : Observation under Antispoofing (may suffer from increased noise)

Bits 0 and 1 for phase only.

Signal strength projected into interval 1-9:
1: minimum possible signal strength
5: threshold for good S/N ratio
9: maximum possible signal strength
0 or blank: not known, don't care

Phase in cycles, Range in meters

04 12 26 0 0	30.0000000 0 9G	4G24G 5G17G 6G10	0G30G 2G29	
-7408143.20348	-5712212.12343	23722895.4574	23722895.8514	23722901.0124
-11151164.34848	-8348759.79145	23027140.6794	23027140.3024	23027147.6974
-17702667.27649	-13496720.20047	21946318.4604	21946318.0704	21946325.1504
-20607717.25049	-16031193.33649	20980332.7214	20980332.1484	20980339.2334
-10697009.82948	-8319281.13543	23671597.2204	23671597.2244	23671604.0324
-25994074.45749	-20224979.69249	20080903.8494	20080902.8804	20080910.1054
-17497598.39549	-13604851.76347	21641129.8624	21641129.7384	21641136.2574
-24900942.06749	-19353992.61648	20874424.4194	20874423.9874	20874428.6824
-2640345.03446	-1780147.16442	24402022.2324	24402021.1924	24402029.2134

C1

11

L2

P1

P2

56

Input format is "fortranny" (Hollerith) (fixed number of digits per data entry field, in fixed "card columns", can leave field blank for zero or no data)

Plus more for

Navigation

"met" (METEOROLOGICAL)

Tílt

Other?